This document is guaranteed to be current only to issue date.

Some Mars Global Surveyor documents that relate to flight operations are under revision to accommodate the recently modified mission plan.

Documents that describe the attributes of the MGS spacecraft are generally up-to-date.

# Mars Global Surveyor

**Mission Operations Specifications** 

Volume 5: Interfaces

Part 1: Software Interface Specification

**Final** 

January 26, 1996



JPL D-12369, Volume 5, Part 1

# MARS GLOBAL SURVEYOR

# MISSION OPERATIONS SPECIFICATION

# **VOLUME 5**

Part 1: Software Interface Specifications

# **Final**

January 26, 1996



JPL D-12369, Vol. 5

# MARS GLOBAL SURVEYOR

# MISSION OPERATIONS SPECIFICATION

# **VOLUME 5**

Part 1: Software Interface Specifications

Prepared:

Richard D Benson, GDS Software System Engineer (acting)

Mars Global Surveyor Project

Approved

F. Hammer GDS Lead Engineer Mars Global Surveyor Project

Jet Propulsion Laboratory California Institute of Technology

JPL D-12369, Vol. 5

# 542-409, Volume 5, Part 1

# DOCUMENT CHANGE LOG

Effective Date	Version	Incorporated MCRs	Affected Sections
01/26/96	Original	N/A	All
02/14/96		(01), (012)	Document Change Log - added to the Original document (to be inserted following signature page);
			Table of Contents - updated to reflect MGS SIS EAE003 Rev. A supplement;
			MO SIS EAE003 dated 01/24/91 revised and reissued as MGS SIS EAE003 Rev A dated 01/25/96, which is incorporated into and supplements this document

### DISTRIBUTION:

M H Acuna (MAG)

Joe Beerer

Rich Benson

Joy Bottenfield

**Bob Brooks** 

Allan Bucher (LMA-Denver)

Mick Connally

P R Christensen (TES)

Sam Dallas

Stuart Demcak

John Ekelund

Pat Esposito

Fred Hammer

Julia Henricks

Bill Hyland

Jeff Jones

M C Malin (MOC)

Jim Newman (LMA-Denver)

Steve Odiorne (LMA-Denver)

Frank Singleton

D E Smith (MOLA)

Richard Southern

Tom Thorpe (2 copies)

G L Tyler (RS) Dave Wagner

Steve Wissler

# 542-409, Vol. 5, Part 1

# Table of Contents

1	INTRODUCTION
	1.1 PURPOSE AND SCOPE
	1.2 INTERFACES DESCRIPTION
	1.3 RESPONSIBILITIES
	1.4 CHANGE CONTROL 4
	1.5 INTERFACE VALIDATION
	1.6 SIS DEFINITION
	1.7 SOFTWARE INTERFACES LISTING
	APPENDIX A
2	SOFTWARE INTERFACE SPECIFICATIONS
	2.1 EAE003 Rev A dated 01/25/96

### Section 1

### Introduction

### 1.1 PURPOSE AND SCOPE

Volume 5, Part 1 specifies software interfaces of Mars Global Surveyor (MGS) Mission Operations System (MOS). It is the controlling document of MGS MOS interfaces for the duration of the Project.

### 1.2 INTERFACES DESCRIPTION

Software interfaces are described in the form of Software Interface Specification (SIS). A SIS covers interfaces between hardware/software components. All identified interfaces are listed in Section 1.7. Each SIS includes a signature page, a brief summary, and a detailed specification of the software interface (as outlined in Appendix A).

Interfaces in which a user team simply accepts and reads a file are defined in Volume 5, Part 2, Operational Interface Agreements.

### 1.3 RESPONSIBILITIES

The design and negotiation of each interface is the responsibility of the Cognizant Interface Engineer; generally the cognizant engineer of the element which produces the product unless otherwise negotiated. The responsibility embraces generation of the agreements, successful negotiation among all parties to these agreements culminating in signed agreements, and renegotiation required as the result of a change proposal on behalf of any party to the agreements. Further, for interfaces defined by reference to existing documentation, the responsible organization of the cognizant interface engineer will monitor the referenced documentation and assure that it is maintained by the cognizant institution and updated consistently with Project requirements.

SISs require the signatures of both generating element engineer(s) and receiving engineer(s) for each affected interface, and the signature of the cognizant interface engineer and the Ground Data System Software System Engineer.

### 1.4 CHANGE CONTROL

This document is subject to the MGS Project Change Control as documented in the MGS Mission System Configuration Management Plan. Any change to any interface agreement included herein requires Mission System Change Control Board approval. Interfaces which are specified by reference to a JPL institutional document will be controlled by Project participation on the institutional change control boards having cognizance over those interfaces.

### 1.5 INTERFACE VALIDATION

The interface validation process is addressed in the Vol. 6, Testing Documents, included therein is a matrix that describes the test cases and validation status of each interface defined in this document.

### 1.6 SIS DEFINITION

A SIS covers interfaces between hardware/software components. Four possible situations may appear:

- 1. A SIS will be written by another organization (external SIS) such as multimission DSN or MGSO specifications which "completely" covers all information required by the project: Cognizant Interface Engineer needs to write only signature page, summary page and PDB information in addition to attaching external SIS.
- 2. More than one external SIS covers all required information: In addition to the signature page, PDB information and summary page, the Cognizant Interface Engineer needs to identify how these SISs are related to each other for this MGS interface. External SIS's need to be included.
- 3. External SIS's exist, but do not "completely" cover all information required by the project: A SIS following the attached SIS outline and written by the Cognizant Interface Engineer is required; however, where applicable, a reference to the existing SISs) and their parts should be specified. In addition, the signature page and PDB information shall be provided.
- 4. An external SIS does not exist: A SIS following the attached SIS outline, a signature page, PDB information and a summary page should be written and prepared by the Cognizant Interface Engineer.

### 1.7 SOFTWARE INTERFACES LISTING

Table 1 is a list of the software interfaces active for MGS. Figure 1 provides a graphical representation of the GDS internal interfaces.

In many cases, the MGS Project SIS is simply an endorsement of a multimission specification. Deep Space Network specifications are contained in document 820-13; Rev A, DSN System Requirements, Detailed Interface Design. You can access the DSN interfaces electronically at the DSN Interface Server, http://jaguar.jpl.nasa.gov/.

Multimission Ground System Organization (MGSO) specifications are contained in the Advanced Multimission Operations System (AMMOS) library. You can access the MGSO interfaces electronically at the MGSO AMMOS Documentation Homepage, http://div390-www/ammos\_docs/.

Note that the original SIS's produced for MO were widely distributed as hardcopy products. (MO document 642-315, Vol. 5, Part 1/JPL D-3822.) In many cases, an electronic copy was not available to MGS personnel. In keeping with MGS policies toward electronic documentation and minimizing cost, SISs which are unchanged from MO are not reissued. These reside in the GDS library and JPL vellum files. As updates are accomplished for MGS, electronic versions will be included on the Project document server.

#### APPENDIX A

### SIS TOPICAL OUTLINE

### 1.0 Introduction

- 1.1 Purpose
- 1.2 Scope
- 1.3 Applicable Documents
- 1.4 Functional Description
  - 1.4.1 Data Content Summary
  - 1.4.2 Source and Transfer Method

Describes where this interfaces is created, and how it is transferred to recipient(s).

1.4.3 Recipients and Utilization

Describes who will receive this interface and what their utilizations are.

1.4.4 Pertinent Relationships with other Interfaces

Describes, if any, other SIS's needed to interpret this interface.

### 1.5 Assumption and Constraints

If there is any assumption(s) made in the generation of this document or file, it should be stated here. Also, if there are any usage constraints, or generation constraints, they should be stated here.

### 2.0 ENVIRONMENT

### 2.1 Hardware Characteristics and Limitation

This paragraph describes any information related to Hardware/Software in order to use this interface.

- 2.1.1 Special Equipment & Device Interfaces
- 2.1.2 Special Setup Requirements

# 2.2 Interface Medium Characteristics

Describe characteristics of the interface media. If the interface media is the PDB, describes the key words of this interface used in the PDB catalog.

- 2.3 Failure Protection, Detection and Recovery
- 2.4 End-of-File convention

### 3.0 ACCESS

#### 3.1 Access Tools

Describe particular programs, utilities, tools needed to access the data.

- 3.2 Input/Output Protocols
- 3.3 Timing and Sequencing Characteristics
- 3.4 PDB Information

Describe PDB data type and associated catalog attributes (if applicable).

#### 4.0 DETAILED INTERFACE SPECIFICATIONS

4.1 Labeling and Identification

This paragraph will describe the version, file naming conventions.

### 4.2 Structure and Organization Overview

The overall file structure is described here. For example, the organization of file header and trailer in conjunction with data records (record header if any). The following diagram is an example.

File Header
Record 1 Header
Record 1 Body
Record 2 Header
Record 2 Body
•
File Trailer

# 4.3 Substructure Definition and Format

# 4.3.1 Header/Trailer Description Details

This paragraph should give detailed description of the user-defined headers, trailers and CHDO headers. Other SFDU headers and trailers should be referenced to the "SFDU Envelop SIS", to be produced by the Data Storage and Retrieval Element (DSRE) of the Ground Data System.

# 4.3.2 Data Description Details

This paragraph describes the detailed format of the data record in a bit level.

### 4.4 Volume, Size and Frequency Estimates

Describe, by phases:

- 1. the size of a object derived from the sizes of record, headers and trailers,
- 2. the frequency of production,
- 3. approximate number of objects, and
- 4. approximate number of objects required on-line.

# MGS Software Interface Specifications Summary

92 SIS's were referenced in MO Volume 2

30 are deleted as not needed for MGS operations

63 SIS's are on the active list for MGS

40 have no change from Mars Observer

5 are reassigned to Engineering Analysis (EAE019/DAC/014, EAE021/DSR017, EAE022/DAC016, EAE023/DAC017, EAE025/DAC029)

13 are files with updated multimission SIS references

5 are revised for MGS:

EAE003	Rev A	Angular Momentum
EAE006	Rev A	SCT Report
NAE003	Rev B	OPTG
NAE006	Rev B Chg 1	MPF
PSE018	new	SEQ Conditions

<u>I/F ID</u>	<u>I/F TITLE</u>	Cog Engineer	Generating Subsystem(s)	Receiving Subsystem(s)	<u>Status</u>
DACE001	DSN Viewperiod File	SFOC-1-SEG-DSN-	V DAC_NOCC	(DSR) EAS NAS PSS	nc
DACE003	DSN Allocation File	SFOC-1-SEG-DSN-	A DAC_NOCC	(DSR) NAS PSS	nc
DACE004	Archival Tracking Data File	e TRK-2-25	DAC_NOCC	(DSR) SSS NAS	nc
DACE005	Orbit Data File (ODF)	TRK-2-18	DAC_NOCC	(DSR) SSS NAS	nc
DACE006	Media Calib Data File	TRK-2-23	DAC_NOCC	(DSR) SSS NAS	nc
DACE007	Time & Polar Motion File	TRK-2-21	DAC_NOCC	(DSR) SSS NAS	nc
DACE009	Tracking Station Locations and Err File	810-5, GEO-10	DAC_text	(DSR) SSS	nc
DACE011	Raw DSN Monitor Blocks	MON-5-15	DAC_CCP	DAC_TIS (DSR) SSS	nc
DACE012	S/C Channelized Engr Data	SFOC-5-TIS- *DU-MGSSFDU	DAC_TIS	(DSR) EAS SSS DAC	new ref 08/23/95
DACE013	Channelized Monitor Data	SFOC-5-TIS- *DU-MGSSFDU	DAC_TIS	(DSR) DAC NAS EAS SSS	new ref 08/23/95
DACE014	(see EAE019)				
DACE016	(see EAE022)				
DACE017	(see EAE023)				
DACE022	Weather Data	TRK-2-24	DAC_TSAC	(DSR) SSS	nc
DACE025	DSN Formatted Cmd File (CMD_DSN, aka GCMD)	SFOC-1-CMD- MO-GNDCMDFIL	DAC_CMD	(DSR) PSS(r) DAC	nc

542-409, Vol. 5 Part 1 **Table 1. MGS Software Interface Specification List** 

<u>I/F ID</u>	<u>I/F TITLE</u>	Cog Engineer	Generating Subsystem(s)	Receiving Subsystem(s)	<u>Status</u>
DACE029	(see EAE025)				
DACE034	QQC Records	SFOC-2-TIS- ANY-MOTELEM	DAC_TIS	(DSR) SSS DAC	nc
DACE037	Expanded Channel Data Record (ECDR)	SFOC-1-DMD- ANY-EXPCHAN	DAC_DMD	EAS SSS (DSR)	nc
DACE040	S/C ENG Telemetry Packet Data Record	SFOC-5-TIS- *DU-MGSSFDU	DAC_TIS	(DSR) EAS SSS	new ref 08/23/95
DACE043	Raw Telemetry Frames	SFOC-1-GIF- DSN-MOGCFTLM	DAC	DAC (Note: not stored on PDB/TDS)	nc
DACE044	Radio Science Open Loop SCP & SSI Data	RSC-11-12	DAC_DSPR	(DSR) SSS	nc
DACE045	Radio Science Open Loop SCP Data Decom Map	RSC-11-12	DAC_DSPR	(DSR)	nc
DACE046	Radio Science Open Loop Data	RSC-11-11	DAC_DSPR	(DSR) SSS [PDS - via RS Team]	nc
DSR001	Event Kernel Data Record	SFOC-1-NAF- Any-EKernel	DSR_NAIF	(DSR), SSS, DAC_rtot	nc
DSR004	MAG/ER Packet Data Reco	ord	DAC_TIS	SPA, SSS	nc
DSR005	TES Packet Data Record		DAC_TIS	SSS SPA	nc
DSR006	MOC Packet Data Record		DAC_TIS	SSS SPA	nc
DSR007	MOLA Packet Data Record		DAC_TIS	SSS SPA	nc

<u>I/F ID</u>	<u>I/F TITLE</u>	Cog Engineer	Generating Subsystem(s)	Receiving Subsystem	n(s) Status
DSR012	QQC Summary Report File	G Smith	DSR_custom	(DSR) EAS SSS	nc
DSR017	(see EAE021)				
DSR019	SPICE Leapseconds File	SFOC-1-NAF- Any-LSKernel	DSR_NAIF	(DSR)	nc
DSR020	SPICE SCLK File	SFOC-1-NAF- Any-SCLKernel	DSR_NAIF	(DSR)	nc
EAE003	Angular Momentum Desaturation	P.Esposito	EAS_AMDGEN	(DSR) NAS	Rev A 1/25/96
EAE006	SCT System Report File	J. Neuman	EAS_Word 6.0	(DSR) SSS NAS	Rev A 1/26/96
EAE007	C-KERNEL (Reconstructed)	E. Dukes	EAS_attrec	(DSR) SSS NAS	nc
EAE008	Maneuver Performance Data File	P.Esposito	EAS_MGPROP	(DSR) NAS_MOPS	nc
EAE011	NAV Engr Information File	P.Esposito	EAS_ Word 6.0	(DSR) NAS_no app	olic s/w nc
EAE014	Maneuver Implementation File	P.Esposito	EAS_MGPROP	(DSR) NAS_MOPS	nc
EAE019	Telecom Performance Predictions	SFOC-1-TAS- Any-TelcomDat	EAS_TPAP	(DSR) DAC_TOPS	reassigned (formerly DACE014)
EAE020	(same as PSE017)				added 2nd ID for EAS

<u>I/F ID</u>	<u>I/F TITLE</u>	Cog Engineer	Generating Subsystem(s)	Receiving Subsystem(s)	<u>Status</u>
EAE021	Spacecraft Clock Coefficient File	SFOC-1-DPS- Any-SCLKvSCET	EAS_SCLKGEN	N (DSR) PSS SSS DAC (	reassigned formerly DSR017)
EAE022	Engr Channel Parameter Table	O. Short SFOC ref: TBD	EAS_TDE	(DSR) DAC (fo	reassigned rmerly DACE016)
EAE023	Channel Conversion Language (CCL) File	O. Short	EAS_ Word 6.0	(DSR) DAC (for	reassigned merly DACE017)
EAE025	Decom Map	O. Short SFOC-1-TIS- Any-DecomMap	EAS_TDE	(DSR) DAC SSS (form	reassigned merly DACE029)
LUE001	Launch Polynomials	P.Esposito	LUE (change probab	DAC NAS ble pending LVWG negotian	nc ions with MDAC)
LUE003	Injection Initial Conditions	TRK-2-17	LUE	DAC NAS	new ref 2/28/94
MMC004	GDS Status Report	F.Hammer	editor	(DSR)	nc
NAE001	Station Polynomial File	P.Esposito	NAS_DPTRAJ	(DSR) EAS_TPAP	nc
NAE002	Light Time File	P.Esposito	NAS_DPTRAJ	(DSR) DAC_CMD PSS_SEQGEN, SEQTRAN EAS_SCLK_correl SSS	nc
NAE003	Orbit Propagation & Timing Geometry File	P.Esposito	NAS_DPTRAJ	(DSR) PSS_SEQGEN SSS_SEQGEN EAS(r)	Rev B 1/15/96

542-409, Vol. 5 Part 1 **Table 1. MGS Software Interface Specification List** 

<u>I/F ID</u>	<u>I/F TITLE</u>	Cog Engineer	Generating Subsystem(s)	Receiving Subsystem(s)	<u>Status</u>
NAE006	Maneuver Profile File	P.Esposito	NAS_MOPS	(DSR) EAS_MGPROP	Rev B Chg 1 9/25/95
NAE007	Planetary Constants Kernel File (PcK)	SFOC-1-NAF- Any-PcKernel	NAS_NAIF	(DSR) EAS(r) SSS(r)	nc
NAE008	Astrodynamic Constants & Init Cond File	P.Esposito	NAS_DPTRAJ	(DSR) SSS(r) EAS(r)	nc
NAE009	Spacecraft Ephemeris File (a.k.a. P-File)	P.Esposito	NAS_DPTRAJ (Note: does not	DAC_DSNOT treside on PDB)	nc
NAE011	S&P Kernels (SPK)	SFOC-1-NAF- Any-SPKernel	NAS_NAIF	(DSR) EAS_STAR/TRANS SSS_MarsMapper	nc
NAE016	Orbit Number File	P.Esposito	NAS_editor	(DSR)	nc
PSE001	Predicted Events File	SFOC-1-SEQ- ANY-PEF	PSS_SEQGEN	(DSR) DAC_SEG SSS/NAS/EAS_SEQREVIEW	new ref 5/23/95
PSE002	Sequence of Events File	SFOC-1-SEG- Any-SOE	DAC_SEG	(DSR) NAS(r) EAS(r) SSS(r)	new ref 5/28/93
PSE003	Space Flight Operations Schedule File	SFOC-1-SEG- Any-SFOS	DAC_SEG	(DSR) NAS(r) EAS(r) SSS(r)	new ref 5/28/93
PSE004	Spacecraft Command Message File	SFOC-1-SEQ- CMD-SCMF	PSS_SEQTRAN	(DSR) DAC_CMD	new ref 10/28/94
PSE005	DSN Keyword File	SFOC-1-SEG- DSN-KEYWORDS	DAC_SEG	(DSR) DAC_NOCC PSS(r)	nc

542-409, Vol. 5 Part 1 **Table 1. MGS Software Interface Specification List** 

<u>I/F ID</u>	<u>I/F TITLE</u>	Cog Engineer	Generating Subsystem(s)	Receiving Subsystem(s)	<u>Status</u>
PSE008	Cumulative Memory Map	SFOC-3-SEQ- MEMFIL	PSS_ SEQTRAN	N (DSR) EAS_MST	new ref 5/26/95
PSE010	S/C Activity Sequence File	SFOC-1-SEQ- Any-SASF	PSS_SEQGEN	(DSR) PSS EAS SSS _SEQGEN	new ref 5/25/95
PSE011	S/C Sequence File	SFOC-2-SEQ- SEQ-SSF	PSS_ SEQGEN	(DSR) PSS_SEQTRAN	new ref 4/25/94
PSE013	Desired Memory Word File	SFOC-3-SEQ- DMWF	PSS_SEQTRAN	(DSR) EAS_MST	new ref 7/10/95
PSE017	Trigger File (a.k.a. EAE020)	SFOC-1-SEG- Any-Trigger	EAS_editor SSS_editor NAS_editor	(DSR) DAC_SEG_soegen	new ref 8/20/93
PSE018	Conditions	SFOC-3-SEQ- CONDITIONS	PSS_SEQGEN	<u>` ′</u>	ew for MGS HECK FINCON)
SSE001	I-Kernel (IK)	SFOC-1-NAF- Any-IKernel	SSS_NAIF	(DSR) SSS_NAIF	nc
SSE004	Instrument Status Report		SSS_text	(DSR) EAS(r)	nc

# Notes:

The "Cognizant Engineer" field shows the ID of the multimission specification in cases where that is the real information source.

<sup>&</sup>quot;nc" means no change from the Mars Observer (MO) version of the SIS; the MO version stands even though the names have changed.

Generating

<u>I/F ID</u> <u>I/F TITLE</u> <u>Cog Engineer</u> <u>Subsystem(s)</u> <u>Receiving Subsystem(s)</u> <u>Status</u>

"new ref" means the multimission spec has been updated since the MO version. In general, MGS will not produce new cover and signature pages just because of an update in the DSN or SFOC references. However, principals will be notifed of such updates.

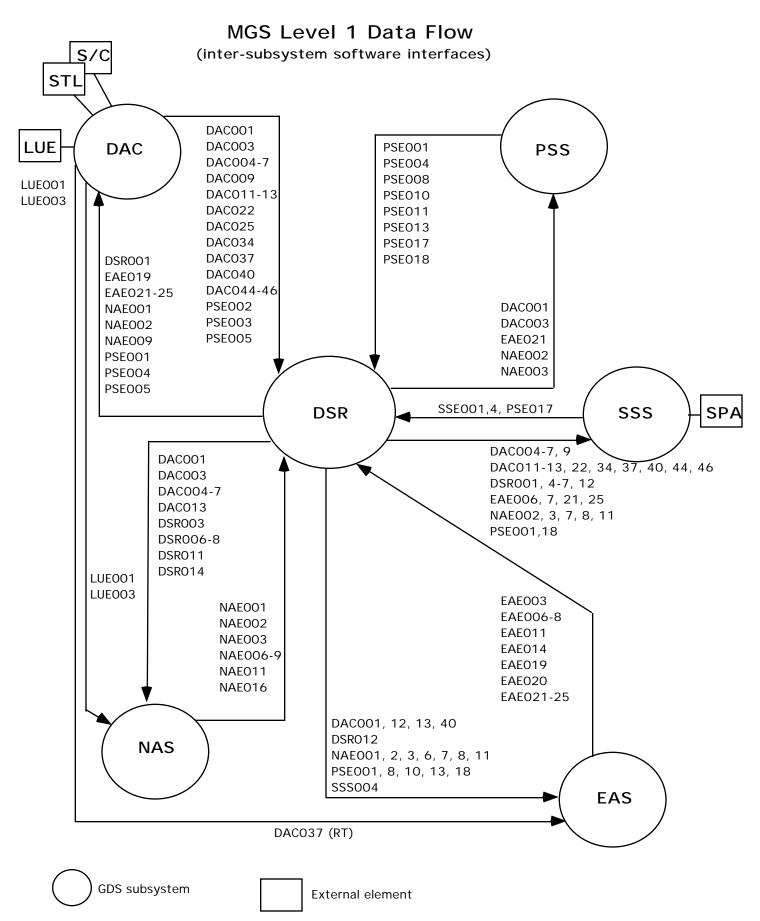
<u>I/F ID</u>	<u>I/F TITLE</u>	Cog Engineer	Generating Subsystem(s)	Receiving Subsystem(s)	<u>Status</u>
Following SIS notify Rich B		find any support for k	eeping these activ	e. Any comments, objections, bet	ter ideas -
DACE002	Command Database		EAS	DAC (DSR) PSS EAS	deleted
DACE020	Command Report [TBD]		DAC	EAS DSR	
DACE031	DSN-GIF MGDS Gateway	,	DAC	—DAC	
DACE041	Command Transmission Protocol	SFOC-1-DSN-CMD CMDMSG	- DAC_CMD	DAC_CPA	delete
DACE042	Command Translation Tabl	e	—DAC	-DAC EAS PSS (DSR)	
dace047	ascii trk dat file	per espo11/21/95			
DACE048	Browser Config. Files		—DAC	SSS, DAC, (DSR)	
DACE049	DMD TDL File		—DAC	SSS, DAC, (DSR)	
DSR011	Raw Data Query Results Fi	le	DSR	EAS SSS	delete
DSR013	Database Catalog Search Results File		DSR	EAS SSS	deleted
DSR015	PDB Objects and Attributes		ALL	— <del>DSR</del>	deleted
DSR016	Inputs to E-KERNEL		PSS, SSS DAC, [EAS]	– <del>DSR</del>	deleted
EAS017	Stream Data Extract Command Line Input IF		EAS SSS	—DSR-	
EAS018	Database Catalog		EAS SSS	— <del>DSR</del>	deleted

<u>I/F ID</u>	<u>I/F TITLE</u>	Cog Engineer	Generating Subsystem(s)	Receiving Subsystem(s)	<u>Status</u>
	Command Line Input I/F				
LUE002	Predicted Injection Error M	atrix P.Esposito	<u>LUE</u>	NAS	delete?
nae004	planet ephem	per esposito 24 oct 9	5		
NAE005	Satellite Ephemeris and Errors File	P.Esposito	NAS	NAS (DSR)	delete?
NAE015	Solar Coronal Model Correction (memorandum)	P.Esposito	DAC	–NAS	delete
PSE014	Spacecraft Checkout Station Command Data File	<u> </u>	PSS	Spacecraft System	delete
SSE005	Instrument Power Profile—	B.Brooks	SSS_text	PSS (DSR) EAS	delete
DSRxx	Telemetry Dictionary	O. Short	SCT	DAC, SSS, PSS, EAS, NAS	cancel
DSR0xx	Command Dictionary	O. Short	SCT	DAC, SSS, PSS, EAS, NAS	cancel
DSR0xx	Block Dictionary	J. Neuman	SCT	DAC, SSS, PSS, EAS, NAS	cancel
DSR021 (No software : Project docur	Flight Rules reads the dictionary files direction are servers eliminates dependent	J. Neuman ctly, SIS proposal was dence on PDB for life	SCT s simply to allow se of mission storage	DAC, SSS, PSS, EAS, NAS storage on the PDB. Commitment ge.)	cancel to maintain
VTL001	VTL Tlm stream		VTL	DAC_cta21 delete - covered by	TTACS
VTL002	VTL Sequence Output		VTL	EAS delete - STL accepts std in	puts
VTL003	VTL Miscomp. Events		VTL	EAS delete - std products used	

542-409, Vol. 5 Part 1 **Table 1. MGS Software Interface Specification List** 

<u>I/F ID</u>	<u>I/F TITLE</u>	Cog Engineer	Generating Subsystem(s)	Receiving Subsystem(s)	<u>Status</u>
VTL004	VTL Memory Dump		VTL	EAS delete - std products used	
VTL005	VTL Sim Runlog		VTL	EAS delete - internal to STL	
VTL006	VTL Tape to Sim		VTL	DAC_Sim delete - covered by T	TACS

Note: The parenthetical (r) indicates a file is received for review only - it is not required by the receiving subsystem.



542-409, Volume 5, Part 1, EAE003, Rev A

# MARS GLOBAL SURVEYOR

Software Interface Specification

# Angular Momentum Desaturation (AMD) File

January 25, 1996



JPL D-12369 Volume 5, Part 1, EAE003, Rev A

# ANGULAR MOMENTUM DESATURATION FILE Software Interface Specification Revision A

25 January 1996

ABSTRACT: This document describes the form and syntax of the Angular Momentum Desaturation File generated by the Engineering Analysis Element (EAE).

National Aeronautics and Space Administration

Jet Propulsion Laboratory California Institute of Technology Pasadena, California 91109

# MARS GLOBAL SURVEYOR SOFTWARE INTERFACE SPECIFICATION

Interface Title: Angular Momentum Desaturation (AMD) File					
Interface ID:	EAE003	Versio	n:	Revision A	
Author:	Revision by S.	original 24 Jan Odiorne 26 Au	1	Kelimo	
Generating Element	Engineer:	Engineering A	nalysis		
/s/ original signe Steve Odiorne	ed by S Odiorne		26 Aug Date	ust 1995	
Receiving Element E	ngineer:	Navigation An	alysis		
P. B. Esposito	uto		2) 14) Date	96	
Receiving Element E	ngineer:	Data Storage a	nd Retrie	val	
John Swift MGS PDB Adn	7		/-23 Date	- 96	
Approval Name and	Title:				
P. B. Esposito Cognizant Intern		: 	2) )4) Date	96	
R. D. Benson GDS Software	System Enginee	r	14 Feb Date	96	

# DISTRIBUTION LIST

Benson, R.D.	264-214
Bottenfield, J.	264-214
Demcak, S. N.	301-150
Ekelund, J. E.	301-220G
Esposito, P. B.	301-150
Hammer, F.	264-214
Jai, B.	301-235
Krug, J.	264-214
Neuman, J.	LMA
Odiorne, S.	LMA
Schmidling, J.	264 225
semmaning, v.	264-235
Sunseri, R. F.	204-235 301-220G
<b>O</b> ,	

# **DOCUMENT CHANGE LOG**

Change Letter	Date	Affected Portions
Original	1/24/91	All
Revision A	1/25/96	All; reissued whole document (deleted & renumber data records to reflect reduced MGS thruster set) [Sign-off complete 2/14/96]

# List of TBD Items

Page Resolution Date Item

25 January 1996 Change Log

# TABLE OF CONTENTS

1.	GENERAL DESCRIPTION	1
1.1	Content Overview	1
1.2	Scope	1
1.3	Applicable Documents	
1.4	Subsystem Siting	1 1
1.4.1	Interface Location, Medium	
1.4.2	Data Source, Destinations, and Transfer Method	2
1.4.3	Generation Method and Frequency	2
1.4.4	Pertinent Relationships with Other Interfaces	1 2 2 2 2 2 2
1.4.5	Labeling and Identification	2
1.5	Assumptions and Constraints	2
2.	INTERFACE CHARACTERISTICS	3
2.1	Hardware Characteristics and Limitations	3
2.1 2.2 2.3	Volume and Size	3 3 3 3 3 3
2.3	Interface Medium Characteristics	3
2.4	Failure Protection, Detection, and Recovery Features	3
2.5	End-of-File (or Medium) Conventions	3
3.	ACCESS	4
3.1	Programs Using the Interface Data	4
3.2	Synchronization Considerations	4
3.3	Input/Output Protocols, Calling Sequences	4
4.	DETAILED INTERFACE SPECIFICATIONS	5
4.1	Structure and Organization Overview	5 5 5 5 5
4.2	Data Format and Definition	5
4.2.1	SFDU Data Description	5
4.2.2	Non-SFDU Data Description	5
4.2.2.1	Header Record 1	6
4.2.2.2	Header Record 2	6
4.2.2.3	Header Record 3	6
4.2.2.4	Data Record 1	6
4.2.2.5	Data Record 2	6 7 7
4.2.2.6	Data Record 3	7
4.2.2.7	Data Record 4	8
4.2.2.8	Data Record 5	8 8 9
4.2.2.9	Data Record 6	
4.2.2.10	Data Record 7	10
4.2.2.11	Data Record 8	10
4.2.2.12	Data Record 9	11
4.2.2.13	Data Record 10	12

25 January 1996 Table of Contents

# Angular Momentum Desaturation (AMD) File

<b>A</b> .	SAMPLE WRAPPED FILE	17
4.2.2.20	End of File Record	16
4.2.2.19	Data Record 16	16
4.2.2.18	Data Record 15	15
4.2.2.17	Data Record 14	14
4.2.2.16	Data Record 13	14
4.2.2.15	Data Record 12	13
4.2.2.14	Data Record 11	12

25 January 1996 Table of Contents

# 1. GENERAL DESCRIPTION

# 1.1. Content Overview

This Software Interface Specification (SIS) contains the description of the Angular Momentum Desaturation (AMD) File.

# 1.2. Scope

This document describes information pertaining to the spacecraft's hydrazine thrusters used in the unloading of momentum stored by the spacecraft's reaction wheels.

The format and syntax specifications in this SIS apply to all phases of the mission.

# 1.3. Applicable Documents

SFOC0038-05-25-01	10 Jan 94	SFOC-1-CDB-ANY-CATALOGUE2
SFOC009-01-00	13 Jun 90	Data Dictionary
SFOC0038-01-14-01	4 Oct 90	SFOC-1-CDB-MGN-TIMESFILE
SFOC0038-02-27-05	30 Aug 94	SFOC-2-SYS-ANY-TIMEFORMS
	1 Jul 90	DPTRAJ-ODP User Reference Manual, SOM Vol. 1 and Vol. 2
	1 Jul 90	DPTRAJ and ODP Interfaces and File Format Descriptions, SOM, Vol. 3

### 1.4. Subsystem Siting

# 1.4.1. Interface Location, Medium

This interface shall consist of a file generated in formatted FORTRAN.

# 1.4.2. Data Source, Destinations, and Transfer Method

The AMD File is generated on the EAE computer. The destination of the data is DPTRAJ within the NAE.

# 1.4.3. Generation Method and Frequency

The AMD file will be generated as required to support NAV operations. Angular Momentum Desaturations (AMDs) are autonomously activated.

# 1.4.4. Pertinent Relationships with Other Interfaces

None.

# 1.4.5. Labeling and Identification

Internal labeling of the AMD File is accomplished with header records as defined in Section 4.2.

# 1.5. Assumptions and Constraints

N/A

# 2. INTERFACE CHARACTERISTICS

# 2.1. Hardware Characteristics and Limitations

N/A

### 2.2. Volume and Size

This file will consist of 80 byte records consisting of 3 AMD header records and 16 data records. Other data will be the SFDU header and file trailer records.

### 2.3. Interface Medium Characteristics

The interface medium will be a computer disk file written in ASCII format.

# 2.4. Failure Protection, Detection, and Recovery Features

None.

# 2.5. End-of-File (or Medium) Conventions

\$\$EOF shall designate end-of-file.

# 3. ACCESS

# 3.1. Programs Using the Interface Data

See cover sheet.

# 3.2. Synchronization Considerations

N/A

# 3.3. Input/Output Protocols, Calling Sequences

N/A

25 January 1996 ACCESS

# 4. DETAILED INTERFACE SPECIFICATIONS

# 4.1. Structure and Organization Overview

The Angular Momentum Desaturation File consists of the SFDU header, the AMD headers, and the AMD data records. Each will be described in the following sections.

### 4.2. Data Format and Definition

# 4.2.1. SFDU Data Description

The SFDU information and structure for the AMD File, in diagrammatic representation:

```
CCSD3ZS00001AAAAAAAA

NJPL3KSOLO15BBBBBBBB

<CR><LF>
MISSION_NAME=MARS_GLOBAL_SURVEYOR; <CR><LF>
SPACECRAFT_NAME=MARS_GLOBAL_SURVEYOR; <CR><LF>
DATA_SET_ID=ANGULAR_MOMENTUM_DESAT; <CR><LF>
FILE_NAME=amdfile.x; <CR><LF>
PRODUCER_ID=EAS; <CR><LF>
APPLICABLE_START_TIME=1993-001T00:00:00.000; <CR><LF>
APPLICABLE_STOP_TIME=1993-001T15:38:31.800; <CR><LF>
PRODUCT_CREATION_TIME-1990-12-21T10:43:000; <CR><LF>
CCSD3RE00000BBBBBBBB
NJPL3IS00255CCCCCCCC
```

In the actual AMD File, the following SFDU data is located at the end of the file, following the non-SFDU data, which is described in Section 4.2.2

# 4.2.2 Non-SFDU Data Description

The non-SFDU portion of the file is described as follows:

#### 4.2.2.1. Header Record 1

Header record 1 identifies the file name.

COLUMNS	CONTENT	DESCRIPTION
1-2	"* "	Header character
3-15	"ANGULARMOMENT"	File identifier (general)
16-27	"HISTORY DATA"	File identifier (specific)

### 4.2.2.2. Header Record 2

Header Record 2 specifies the date and time of file creation.

COLUMNS	CONTENT	DESCRIPTION
1-2	"* "	Header character
3-16	"CREATION: SCT "	Specifies that the file creation is by
		the SCT
17-27	dd-mmm-yyyy	Day, month and year of file creation,
		eg 04-DEC-1989 (local date)
28-28	" "	Blank space
29-36	hh:mm:ss	Hours, minutes and seconds of file
		creation (local time)

## 4.2.2.3. Header Record 3

Header Record 3 contains the user input file description.

i.e., user free to put in his o	COLUMNS	CONTENT	DESCRIPTION
i.e., user free to put in his o	1-2	"* "	Header character
· · · · · · · · · · · · · · · · · · ·	3-80	User input	File description input by user
information here			i.e., user free to put in his own
information here			information here

## 4.2.2.4. Data Record 1

Data Record 1 specifies momentum unload start time.

COLUMNS	CONTENT	DESCRIPTION
1-12	"STRT_TIME = "	Closest indication to when momentum begins to be imparted to the spacecraft.
13-31	yy-ddd/hh:mm:ss.fff	Unload event start time at the spacecraft (UTC).

### 4.2.2.5. Data Record 2

Data Record 2 specifies momentum unload end time.

COLUMNS	CONTENT	DESCRIPTION
1-12	"STOP_TIME = "	Closest indication to when momentum
13-31	yy-ddd/hh:mm:ss.fff	ceases to be imparted to the spacecraft. Unload event stop time at the
	<i>J J</i>	spacecraft (UTC).

## 4.2.2.6. Data Record 3

Data record 3 through 14 represent data for each of the 12 possible thrusters that are numbered 01 through 12. This data record is for thruster number 01.

COLUMNS	CONTENT	DESCRIPTION
1-11	"THRUSTER 01"	Record label. This record is always present, even if thruster is inactive.
12-12	" "	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	I3	Integer number of pulses for thruster $01$ . If $= 0$ , this thruster is inactive. If not $= 0$ , thruster is active.
27-27	" "	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 01.
37-41	F5.2	Time in seconds between pulses.
42-42	11 11	Blank Space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for thruster 01.
54-58	F5.3	Duration or width of the pulse of the active thruster in seconds.
59-59	" "	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pule for thruster 01.
72-79	F8.5	Value of the impulse/pulse in "Newton seconds" for active thruster 01.

## 4.2.2.7. Data Record 4

This data record is for thruster number 02.

COLUMNS	CONTENT	DESCRIPTION
1-11	'THRUSTER 02"	Record label. This record is always present, even if thruster is inactive.
12-12	11 11	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	I3	Integer number of pulses for thruster 02. If = 0, this thruster is inactive. If not = 0, thruster is active.
27-27	11 11	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 02.
37-41	F5.2	Time in seconds between pulses.
42-42	11 11	Blank space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for thruster 02.
54-58	F5.3	Duration or width of the pulse of the active thruster in seconds.
59-59	11 11	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pulse for thruster 02.
72-79	F8.5	Value of the impulse/pulse in "Newton seconds" for active thruster 02.

## 4.2.2.8. Data Record 5

This data record for thruster number 03.

COLUMNS	CONTENT	DESCRIPTION
1-11	'THRUSTER 03"	Record label. This record is always present, even if thruster is inactive.
12-12	" "	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	I3	Integer number of pulses for thruster $03$ . If $= 0$ , this thruster is inactive. If not $= 0$ , thruster is active.
27-27	" "	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 03.

37-41	F5.2	Time in seconds between pulses.
37-41		Time in seconds between pulses.
42-42	" "	Blank space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for
		thruster 03.
54-58	F5.3	Duration or width of the pulse of the active
		thruster in seconds.
59-59	" "	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pulse for
		thruster 03.
72-79	F8.5	Value of the impulse/pulse in "Newton
		seconds" for active thruster 03.

## 4.2.2.9. Data Record 6

This data record is for thruster number 04.

COLUMNS	CONTENT	DESCRIPTION
1-11	'THRUSTER 04"	Record label. This record is always present, even if thruster is inactive.
12-12	11 11	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	I3	Integer number of pulses for thruster 04. If = 0, this thruster is inactive. If not = 0, thruster is active.
27-27	11 11	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 04.
37-41	F5.2	Time in seconds between pulses.
42-42	11 11	Blank space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for thruster 04.
54-58	F5.3	Duration or width of the pulse of the active thruster in seconds.
59-59	11 11	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pulse for thruster 04.
72-79	F8.5	Value of the impulse/pulse in "Newton seconds" for active thruster 04.

## 4.2.2.10. Data Record 7

This data record is for thruster number 05.

COLUMNS	CONTENT	DESCRIPTION
1-11	'THRUSTER 05"	Record label. This record is always present, even if thruster is inactive.
12-12	11 11	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	I3	Integer number of pulses for thruster 05. If $= 0$ , this thruster is inactive. If not $= 0$ , thruster is active.
27-27	11 11	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 05.
37-41	F5.2	Time in seconds between pulses.
42-42	11 11	Blank space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for thruster 05.
54-58	F5.3	Duration or width of the pulse of the active thruster in seconds.
59-59	11 11	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pulse for thruster 05.
72-79	F8.5	Value of the impulse/pulse in "Newton seconds" for active thruster 05.

### 4.2.2.11. Data Record 8

This data record is for thruster number 06.

COLUMNS	CONTENT	DESCRIPTION
1-11	'THRUSTER 06"	Record label. This record is always present, even if thruster is inactive.
12-12	" "	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	13	Integer number of pulses for thruster 06. If $= 0$ , this thruster is inactive. If not $= 0$ , thruster is active.
27-27	" "	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 06.

37-41	F5.2	Time in seconds between pulses.
42-42	" "	Blank space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for
		thruster 06.
54-58	F5.3	Duration or width of the pulse of the active
		thruster in seconds.
59-59	п п	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pulse for
		thruster 06.
72-79	F8.5	Value of the impulse/pulse in "Newton
		seconds" for active thruster 06.

## 4.2.2.12. Data Record 9

This data record is for thruster number 07.

COLUMNS	CONTENT	DESCRIPTION
1-11	'THRUSTER 07"	Record label. This record is always present, even if thruster is inactive.
12-12	11 11	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	13	Integer number of pulses for thruster 07. If = 0, this thruster is inactive. If not = 0, thruster is active.
27-27	" "	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 07.
37-41	F5.2	Time in seconds between pulses.
42-42	" "	Blank space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for thruster 07.
54-58	F5.3	Duration or width of the pulse of the active thruster in seconds.
59-59	11 11	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pulse for thruster 07.
72-79	F8.5	Value of the impulse/pulse in "Newton seconds" for active thruster 07.

## 4.2.2.13. Data Record 10

This data record is for thruster number 08.

COLUMNS	CONTENT	DESCRIPTION
1-11	'THRUSTER 08"	Record label. This record is always present, even if thruster is inactive.
12-12	11 11	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	I3	Integer number of pulses for thruster $08$ . If $= 0$ , this thruster is inactive. If not $= 0$ , thruster is active.
27-27	" "	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 08.
37-41	F5.2	Time in seconds between pulses.
42-42		Blank space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for thruster 08.
54-58	F5.3	Duration or width of the pulse of the active thruster in seconds.
59-59	" "	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pulse for thruster 08.
72-79	F8.5	Value of the impulse/pulse in "Newton seconds" for active thruster 08.

## 4.2.2.14. Data Record 11

This data record is for thruster number 09.

COLUMNS	CONTENT	DESCRIPTION
1-11	'THRUSTER 09"	Record label. This record is always present, even if thruster is inactive.
12-12	" "	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	I3	Integer number of pulses for thruster 09. If $= 0$ , this thruster is inactive. If not $= 0$ , thruster is active.
27-27	" "	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 09.

37-41	F5.2	Time in seconds between pulses.
42-42	" "	Blank space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for thruster 09.
54-58	F5.3	Duration or width of the pulse of the active thruster in seconds.
59-59	н н	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pulse for thruster 09.
72-79	F8.5	Value of the impulse/pulse in "Newton seconds" for active thruster 09.

## 4.2.2.15. Data Record 12

This data record is for thruster number 10.

COLUMNS	CONTENT	DESCRIPTION
1-11	'THRUSTER 10"	Record label. This record is always present, even if thruster is inactive.
12-12	" "	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	I3	Integer number of pulses for thruster 10. If $= 0$ , this thruster is inactive. If not $= 0$ , thruster is active.
27-27	" "	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 10.
37-41	F5.2	Time in seconds between pulses.
42-42	" "	Blank space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for thruster 10.
54-58	F5.3	Duration or width of the pulse of the active thruster in seconds.
59-59	" "	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pulse for thruster 10.
72-79	F8.5	Value of the impulse/pulse in "Newton seconds" for active thruster 10.

## 4.2.2.16. Data Record 13

This data record is for thruster number 11.

COLUMNS	CONTENT	DESCRIPTION
1-11	'THRUSTER 11"	Record label. This record is always present, even if thruster is inactive.
12-12	" "	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	I3	Integer number of pulses for thruster 11. If $= 0$ , this thruster is inactive. If not $= 0$ , thruster is active.
27-27	" "	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 11.
37-41	F5.2	Time in seconds between pulses.
42-42		Blank space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for thruster 11.
54-58	F5.3	Duration or width of the pulse of the active thruster in seconds.
59-59	" "	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pulse for thruster 11.
72-79	F8.5	Value of the impulse/pulse in "Newton seconds" for active thruster 11.

## 4.2.2.17. Data Record 14

This data record is for thruster number 12.

COLUMNS	CONTENT	DESCRIPTION
1-11	'THRUSTER 12"	Record label. This record is always present, even if thruster is inactive.
12-12	" "	Blank space
13-23	"NOPULSES="	Label to indicate the number of pulses fired during the course of a single desaturation event.
24-26	I3	Integer number of pulses for thruster 12. If $= 0$ , this thruster is inactive. If not $= 0$ , thruster is active.
27-27	" "	Blank space
28-36	"DEADTIME="	Label to indicate time between pulses for thruster 12.

37-41	F5.2	Time in seconds between pulses.
42-42	" "	Blank space
43-53	"PULSEWIDTH="	Label of the width or span of the pulse for thruster 12.
54-58	F5.3	Duration or width of the pulse of the active thruster in seconds.
59-59	" "	Blank space
60-71	"IMPULSE BIT="	Label specifying the impulse per pulse for thruster 12.
72-79	F8.5	Value of the impulse/pulse in "Newton seconds" for active thruster 12.

### 4.2.2.18. Data Record 15

Data record 15 specifies the quaternion relating spacecraft axes to J2000 coordinates at the event start time. This quaternion represents the spacecraft orientation at the start of the desaturations.

COLUMNS	CONTENT	DESCRIPTION
1-11	"QUATERNION="	Record label
12-12	" "	Blank space
13-21	F9.6	Value of 1st quaternion component = a.
22-22	" "	Blank space
23-31	F9.6	Value of 2nd quaternion component = b.
32-32	" "	Blank space
33-41	F9.6	Value of 3rd quaternion component = $c$ .
42-42	" "	Blank space
43-51	F9.6	Value of 4th quaternion component $=$ d.

## Where

 $a = e_1 \sin(\phi/2)$ ,

 $b = e_2 \sin(\phi/2),$ 

 $c = e_3 sin(\phi/2)$ , and

 $d = cost\phi/2$ .

### 4.2.2.19. Data Record 16

Data record 16 specifies the spacecraft spin rate at the start of desaturation.

COLUMNS	CONTENT	DESCRIPTION
1-17	"S/C ANGULAR RATE"	Record label.  Angular rate values are applicable to the start of desaturations and are needed for all AACS modes, including mapping. The angular rate is broken into 3 components along the body frame or spacecraft axes.
18-18	11 11	Blank space
19-23	"ROLL="	Label representing the spin rate about the spacecraft roll (x) axis.
24-30	F7.4	Value of the spin rate about the s/c roll axis in deg/sec.
31-32	" "	2 blank spaces
33-38	"PITCH="	Label representing the spin rate about the spacecraft pitch (y) axis.
39-45	F7.4	Value of the spin rate about the s/c pitch axis in degrees/sec.
46-47	" "	2 blank spaces
48-51	"YAW="	Label representing the spin rate about the s/c yaw (z) axis.
52-58	F7.4	Value of the spin rate about the s/c yaw axis in deg/sec.

### 4.2.2.20. End of File Record

The End-of-File Record indicates the end of file.

COLUMNS	CONTENT	DESCRIPTION
1-5	"\$\$EOF"	Signifies end of data

NOTE: The three (3) header, sixteen (16) data, and one (1) end of file records may be repeated as many times as there are AMD events to be recorded. See Appendix, Sample Wrapped File, below.

# Appendix A. SAMPLE WRAPPED FILE

A sample wrapped Angular Momentum Desaturation file is provided in order to show the expected format of the data. The user should not code specifically to the real numbers as printed.

```
CCSD3ZS00001AAAAAAAANJPL3KS0L015BBBBBBBB
MISSION NAME=MARS GLOBAL SURVEYOR
SPACECRAFT_NAME=MARS_GLOBAL_SURVEYOR
DATA_SET_ID=ANGULAR_MOMENTUM_DESAT;
FILE_NAME=amdfile.x;
PRODUCER_ID=EAS;
APPLICABLE_START_TIME=1993-001T00:00:00.000;
APPLICABLE STOP TIME=1993-001T15:38:31.800;
PRODUCT_CREATION_TIME=1990-12-21T10;43;00;
CCSD3RE00000BBBBBBBBBBDJPL3IS00255CCCCCCCC
* ANGULARMOMENTHISTORY DATA
* CREATION: SCT 21-DEC-1990 10:43:00
* SAMPLE AMD FILE - OFF SPIN AXIS
STRT TIME = 93-001/00:00:00.000
STOP\_TIME = 93-001/00:04:03.100
THRUSTER 01 NO._PULSES= 31 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 02 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 03 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
                        O DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
THRUSTER 04 NO._PULSES=
0.44000
THRUSTER 05 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 06 NO._PULSES= 31 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 07 NO._PULSES=
                         O DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 08 NO. PULSES=
                         O DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 09 NO._PULSES=
                         O DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
                        0 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
THRUSTER 10 NO._PULSES=
0.26400
THRUSTER 11 NO. PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
THRUSTER 12 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
QUATERNION= 0.250000 0.250000 0.353553 0.866025
S/C ANGULAR RATE ROLL= 0.0000 PITCH=-0.0510 YAW= 0.0000
$$EOF
* ANGULARMOMENTHISTORY DATA
* CREATION: SCT 10-DEC-1990 13:14:33
```

```
* SAMPLE AMD FILE - SPIN AXIS
STRT_TIME = 93-001/03:54:00.000
STOP TIME = 93-001/03:57:02.900
THRUSTER 01 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 02 NO. PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 03 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 04 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 05 NO. PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 06 NO. PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 07 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 08 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 09 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
THRUSTER 10 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
THRUSTER 11 NO._PULSES= 23 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
THRUSTER 12 NO._PULSES= 23 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
OUATERNION= 0.250000 0.250000 0.353553 0.866025
S/C ANGULAR RATE ROLL= 0.0000 PITCH=-0.0510 YAW= 0.0000
$$EOF
* ANGULARMOMENTHISTORY DATA
* CREATION: SCT 10-DEC-1990 13:23:20
* SAMPLE AMD FILE - NEG PITCH UNLOAD
STRT TIME = 93-001/07:48:00.000
STOP\_TIME = 93-001/07:51:27.800
THRUSTER 01 NO. PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 02 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 03 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 04 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 05 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 06 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 07 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 08 NO. PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
THRUSTER 09 NO._PULSES= 26 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
```

```
THRUSTER 10 NO. PULSES= 26 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
THRUSTER 11 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
THRUSTER 12 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
OUATERNION= 0.250000 0.250000 0.353553 0.866025
S/C ANGULAR RATE ROLL= 0.0000 PITCH=-0.0510 YAW= 0.0000
$$EOF
* ANGULARMOMENTHISTORY DATA
* CREATION: SCT 10-DEC-1990 13:40:09
* SAMPLE AMD FILE - YAW/PITCH UNLOAD
STRT_TIME = 93-001/11:42:00.000
STOP TIME = 93-001/11:44:09.700
THRUSTER 01 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 02 NO. PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 03 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 04 NO._PULSES= 17 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 05 NO. PULSES= 17 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 06 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 07 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 08 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 09 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
THRUSTER 10 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
THRUSTER 11 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
0.26400
THRUSTER 12 NO._PULSES= 0 DEADTIME= 8.0 PULSEWIDTH=0.30 IMPULSE BIT=
OUATERNION= 0.250000 0.250000 0.353553 0.866025
S/C ANGULAR RATE ROLL= 0.0000 PITCH=-0.0510 YAW= 0.0000
$$EOF
* ANGULARMOMENTHISTORY DATA
* CREATION: SCT 10-DEC-1990 14:43:03
* SAMPLE AMD FILE - YAW EMERGENCY UNLOAD
STRT TIME = 93-001/15:36:00.000
STOP\_TIME = 93-001/15:38:31.800
THRUSTER 01 NO._PULSES= 0 DEADTIME= 4.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 02 NO._PULSES= 38 DEADTIME= 4.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 03 NO._PULSES= 38 DEADTIME= 4.0 PULSEWIDTH=0.10 IMPULSE BIT=
THRUSTER 04 NO._PULSES= 0 DEADTIME= 4.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
```

```
THRUSTER 05 NO._PULSES= 0 DEADTIME= 4.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 06 NO._PULSES= 0 DEADTIME= 4.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 07 NO._PULSES= 38 DEADTIME= 4.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 08 NO._PULSES= 38 DEADTIME= 4.0 PULSEWIDTH=0.10 IMPULSE BIT=
0.44000
THRUSTER 09 NO._PULSES= 0 DEADTIME= 4.0 PULSEWIDTH=0.50 IMPULSE BIT=
0.44000
THRUSTER 10 NO._PULSES= 0 DEADTIME= 4.0 PULSEWIDTH=0.50 IMPULSE BIT=
0.44000
THRUSTER 11 NO._PULSES= 0 DEADTIME= 4.0 PULSEWIDTH=0.50 IMPULSE BIT=
0.44000
THRUSTER 12 NO._PULSES= 0 DEADTIME= 4.0 PULSEWIDTH=0.50 IMPULSE BIT=
0.44000
QUATERNION= 0.250000 0.250000 0.353553 0.866025
S/C ANGULAR RATE ROLL= 0.0000 PITCH=-0.0510 YAW= 0.0000
$$EOF
CCSD3RE00000CCCCCCCCCSD3RE00000AAAAAAA
```